

Printed Pages : 3



EAE-031

(Following Paper ID and Roll No. to be filled in your Answer Book)

PAPER ID : 148751

Roll No.

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B. Tech.

(SEM. III) (ODD SEM.) THEORY

EXAMINATION, 2014-15

AERODYNAMICS-II

Time : 3 Hours]

[Total Marks : 100

Note : Attempt All five questions.
Choice and Marks are given below.

1 Attempt any two of the following : **2×10=20**(a) Define the formula $Cd_i = \frac{CL^2}{\pi AR}$.

(b) What is Biot and Savart law ? How does it affect lift ?

(c) Derive the formula for turning flight what are V-n diagrams ?

2 Attempt any two of the following : **10×2=20**(a) What is the effect of Aspect Ratio on a finite wing ?
Write down the equation for total drag of a wing.

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[Contd...

- (b) Name the fundamental equations of Prandtl's lifting time theory and hence derive the formula for induced drag coefft.
- (c) Derive the linearised velocity potential for subsonic compressible flows over an aerofoil.
- 3** Attempt any two parts of the following : **2×10=20**
- (a) What is compressibility ? Write down equations for inviscid compressible flow.
- (b) Draw the various forces affecting an airplane during take off. State the formulas.
- (c) What is the effect of location of centre of gravity relative to neutral point on static stability.
- 4** Attempt any two of the following : **2×10=20**
- (a) What is a propeller ? Show an illustration of a propeller thereby showing variation of pitch along a blade. Write down formulas for propeller efficiency and advance ratio.
- (b) Write short note on the following :
- (i) Critical mach no
 - (ii) Aspect Ratio
 - (iii) Dutch Roll
 - (iv) Sweep Back effects.
- (c) Write down the equations of motion for dynamic longitudinal stabilities.
- 5** Attempt any two parts of the following : **10×2=20**
- (a) Briefly describe drag divergence with a sketch.

- (b) Write short notes on the following :
- (i) autorotation and spin
 - (ii) accelerated flights
 - (iii) pitching moment
 - (iv) centre of pressure.
- (c) What is the effect of centre of gravity relative to neutral point on static stability.
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